

# Liquid Film Drain from an Accelerating Tank Wall

JOHN R. O'LOUGHLIN\*

The Boeing Company, New Orleans, La.

## Nomenclature

$g$  = acceleration body force  
 $t$  = time  
 $u$  = the  $x$  component of velocity  
 $x$  = distance measured along the tank wall from the position of the bulk liquid interface at  $t = 0$   
 $y$  = distance measured perpendicular to the tank wall  
 $\delta$  = thickness of draining film  
 $\mu$  = dynamic viscosity of liquid  
 $\rho$  = density of liquid  
 $\rho_v$  = density of vapor

IN many studies of a draining cryogenic tank, such as those of wall heat transfer or tank pressurization, it is important to have a knowledge of the liquid film that clings to the tank wall. Such a film is illustrated in Fig. 1. Several investigators have examined a draining film for various reasons in the past. This work has been both analytical and experimental and is described by van Rossum.<sup>1</sup> It is all for constant  $g$ .

Neglecting inertia terms in the momentum equation, the velocity in such a laminar draining film with no slip at the wall and zero shear at the liquid-vapor interface is given by<sup>2</sup>

$$u = \frac{(\rho - \rho_v)g}{\mu} \left( y\delta - \frac{y^2}{2} \right) \quad (1)$$

The continuity equation written for a slab of incompressible liquid  $dx$  in thickness is

$$\frac{\partial}{\partial x} \int_0^\delta u dy + \frac{\partial \delta}{\partial t} = 0 \quad (2)$$

One should note in passing that mass transfer from the liquid film introduces a nonzero term into the right-hand side of Eq. (2).

Substitution of the velocity distribution from Eq. (1) into the continuity equation yields

$$\frac{(\rho - \rho_v)g}{\mu} \delta^2 \frac{\partial \delta}{\partial x} + \frac{\partial \delta}{\partial t} = 0 \quad (3)$$

Equation (3) is a quasi-linear, partial differential equation. The solution of such equations using the method of characteristics is described by Hildebrand.<sup>3</sup> In this case, application of the method reduces to the solution of the following two ordinary differential equations:

$$\frac{dx}{(\rho - \rho_v)g\delta^2/\mu} = \frac{dt}{1} = \frac{d\delta}{0} \quad (4)$$

The third term of this equation signifies that  $\delta$  is a constant along a characteristic. This fact simplifies the integration of the equation formed by the first and second terms. This equation is

$$dx = \frac{(\rho - \rho_v)g\delta^2}{\mu} dt \quad (5)$$

Integration with the boundary condition  $x = 0$  at  $t = 0$  yields the desired film thickness

$$\delta = \left( \frac{\mu x}{(\rho - \rho_v) \int_0^t g dt} \right)^{1/2} \quad (6)$$

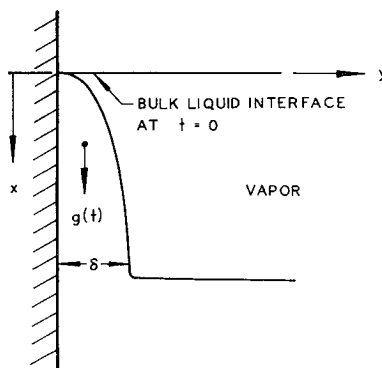


Fig. 1 Physical model.

For a constant  $g$ , Eq. (6) reduces to

$$\delta = \left( \frac{\mu x}{(\rho - \rho_v)g t} \right)^{1/2} \quad (7)$$

This agrees with the result of van Rossum except that he has neglected  $\rho_v$  in comparison to  $\rho$ .

Equation (6) is an interesting result since it reveals that the factor of importance in the film profile is the area under the  $g$  vs time curve. According to this analysis, which neglects surface tension and contact angle, the profile is unchanged during periods of zero  $g$ .

## References

- 1 van Rossum, J. J., "Viscous lifting and the drainage of liquids," *Appl. Sci. Res.* **A7**, 121-144 (1958).
- 2 Sparrow, E. M. and Siegel, R., "Transient film condensation," *J. Appl. Mech.* **81**, 120-121 (1959).
- 3 Hildebrand, F. B., *Advanced Calculus for Applications* (Prentice-Hall, Inc., Englewood Cliffs, N. J., 1963), Chap. 8, pp. 379-389.

## Calculation of Natural Modes of Vibration for Free-Free Structures in Three-Dimensional Space

JULIAN H. BERMAN\* AND JEROME SKLEROV†

Republic Aviation Corporation,  
 Farmingdale, N. Y.

HAVING had occasion to require the calculation of natural free-free modes for large three-dimensional structures, we have devised a procedure for developing the free-free matrix for a structure in three dimensions employing lumped mass and inertia concepts. The extension of this concept to three dimensions was indicated in Ref. 1.

Consider a three-dimensional structure consisting of lumped masses and inertias clamped at some point (the origin). Each of the lumped mass-inertia elements that constitute the structure possesses six degrees of freedom; three translational and three rotational. A matrix of flexibility influence coefficients  $[C]$  exists which relates the six deflections and rotations of each lumped mass-inertia to forces and moments applied at every lumped mass of the structure. A right-hand coordinate system is employed.

The eigenvalue problem to be solved for the cantilevered modes of the structure can be written as

$$\{\delta\} = \omega^2 [C] [M] \{\delta\} \quad (1)$$

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\* Consultant, Launch Systems Branch; also Associate Professor of Mechanical Engineering, Tulane University, New Orleans, La. Member AIAA.

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\* Staff Engineer. Member AIAA.

† Senior Dynamics Engineer.